

Discriminating a Gravitational Wave Background from Instrumental Noise using Time-Delay Interferometry

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Reference:

- [1] M.Tinto, *Phys. Rev. D*: **58**, 102001 (1998)
- [2] M. Tinto & J.W. Armstrong, *Phys. Rev. D*: **59**, 102003, (1999)
- [3] J.W. Armstrong, F.B. Estabrook, & M. Tinto, *Ap. J.*, 527: 814-826, 1999 December 20
- [4] F.B. Estabrook, M. Tinto, & J.W. Armstrong, *Phys. Rev. D*: In press.
- [5] M. Tinto, J.W. Armstrong, & F.B. Estabrook, In Preparation.

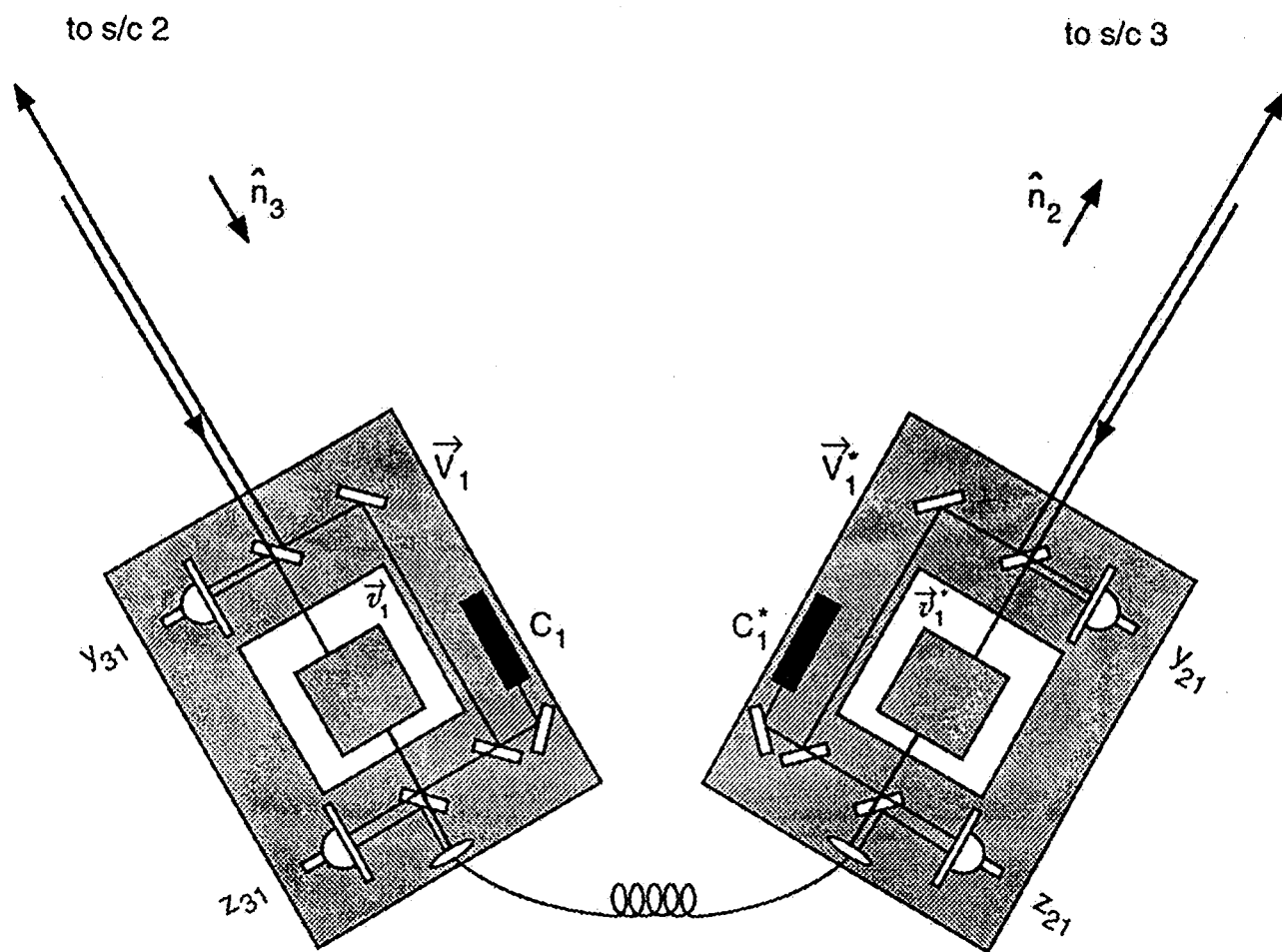
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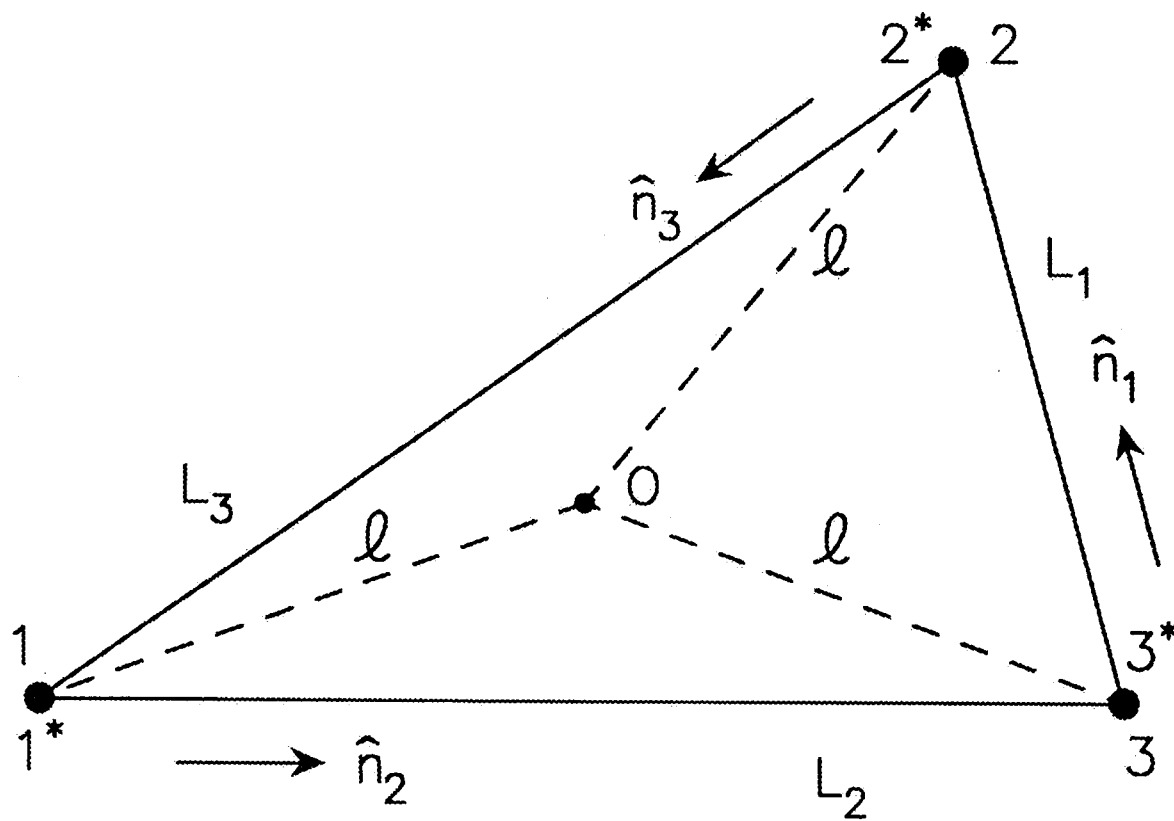
EARTH vs SPACE BASED **INTERFEROMETERS**

- Earth-based interferometers have arm-lengths essentially equal. This is in order to directly remove laser frequency fluctuations at the photodetector, where the two beams interfere.
- Unlike ground-based detectors, LISA can not maintain precise arm-length equality.
- Time-of-flight of laser signals and gravitational waves throughout the apparatus are important ($\lambda_{\text{GW}} \leq L$)
- Spacecraft cannot be isolated, but incorporate drag-free proof masses to allow monitoring of spacecraft accelerations.

Time-Delay Interferometry

- In our previous work [3] we treated time-delay gravitational wave interferometry with three spacecraft, each idealized as moving almost inertially and rigidly carrying optical benches.
- Each spacecraft transmits laser signals to the other two, and using its laser as a local oscillator, measures the frequencies of the laser beams received from the other two.
- In [2, 3] we presented and analyzed combinations of these data streams which eliminated the laser frequency fluctuations (the dominant noise source!), while retaining the gravitational wave signal.
- We have generalized the results derived in [3] to the actual drag-free configuration envisaged for LISA.





Time-Delay Interferometry (Cont.)

- In summary: there are 6 optical benches, 6 lasers, and a total of 12 Doppler time series observed.
- The 6 beams exchanged between distant spacecraft contain the information about the GW signal (y_{ij}); the other 6 signals (z_{ij}) are for comparison of the lasers and relative optical bench motions within the spacecraft.

One-Way Responses

- $y_{21}(t)$ = Doppler measured at spacecraft # 1, with transmission at spacecraft # 3; $y_{21,3} = y_{21}(t - L_3)$
- $z_{21}(t)$ = Doppler measured at bench # 1*, with transmission at bench # 1; $z_{31}(t)$ = Doppler measured at bench # 1, with transmission at bench # 1*.

$$y_{21}(t) = C_{3,2} - \vec{n}_2 \cdot \vec{V}_{3,2} + 2 \vec{n}_2 \cdot \vec{v}_1^* - \vec{n}_2 \cdot \vec{V}_1^* - C_1^* + y_{21}^{\text{gw}}(t) + y_{21}^{\text{shot}}(t)$$

$$z_{21}(t) = C_1 + 2 \vec{n}_3 \cdot (\vec{v}_1 - \vec{V}_1) + \rho_1 - C_1^*$$

$$y_{31}(t) = C_{2,3}^* + \vec{n}_3 \cdot \vec{V}_{2,3}^* - 2 \vec{n}_3 \cdot \vec{v}_1 + \vec{n}_3 \cdot \vec{V}_1 - C_1 + y_{31}^{\text{gw}}(t) + y_{31}^{\text{shot}}(t)$$

$$z_{31}(t) = C_1^* - 2 \vec{n}_2 \cdot (\vec{v}_1^* - \vec{V}_1^*) + \rho_1 - C_1$$

- Eight other relations are obtained by cyclic permutation of the indices in the equations above.

Data Combinations which Eliminate Laser noise and Optical Bench Motions

- Unequal-arm Interferometer response: X

$$\begin{aligned}
 X = & y_{32,322} - y_{23,233} + y_{31,22} - y_{21,33} + y_{23,2} - y_{32,3} + y_{21} - y_{31} \\
 & + 1/2 [-z_{21,2233} + z_{21,33} + z_{21,22} - z_{21}] \\
 & + 1/2 [+z_{31,2233} - z_{31,33} - z_{31,22} + z_{31}]
 \end{aligned}$$

- The remaining noise terms due to the proof-mass motion, \vec{v}_i , \vec{v}_i^* , are:

$$\begin{aligned}
 X^{\text{proof mass}} = & \vec{n}_2 \cdot (-\vec{v}_{1,2233}^* + \vec{v}_{1,22}^* - \vec{v}_{1,33}^* + \vec{v}_1^* + 2\vec{v}_{3,233} - 2\vec{v}_{3,2}) \\
 & + \vec{n}_3 \cdot (-\vec{v}_{1,2233} - \vec{v}_{1,22} + \vec{v}_{1,33} + \vec{v}_1 + 2\vec{v}_{2,223}^* - 2\vec{v}_{2,3}^*)
 \end{aligned}$$

Data Combinations which Eliminate Laser noise and Optical Bench Motions (Cont.)

- Sagnac Interferometer: ξ

$$\begin{aligned}\xi = & y_{32,2} - y_{23,3} + y_{13,3} - y_{31,1} + y_{21,1} - y_{12,2} \\ & + 1/2 [-z_{13,21} + z_{23,12} - z_{21,23} + z_{31,23} - z_{32,13} + z_{12,13}] \\ & + 1/2 [-z_{32,2} + z_{12,2} - z_{13,3} + z_{23,3} - z_{21,1} + z_{31,1}]\end{aligned}$$

- The remaining noise terms due to the proof-mass motion, v_i, v_i^* , are:

$$\begin{aligned}\xi^{\text{proof mass}} = & \vec{n}_1 \cdot (\vec{v}_{2,2} - \vec{v}_{2,13} + \vec{v}_{3,3}^* - \vec{v}_{3,21}^*) \\ & + \vec{n}_2 \cdot (\vec{v}_{3,3} - \vec{v}_{3,21} + \vec{v}_{1,1}^* - \vec{v}_{1,23}^*) \\ & + \vec{n}_3 \cdot (\vec{v}_{1,1} - \vec{v}_{1,23} + \vec{v}_{2,2}^* - \vec{v}_{2,13}^*)\end{aligned}$$

Conclusions

- We have shown that there exist several linear combinations of the twelve data that minimize the gravitational wave signal.
- In the frequency region of interest they all display the same sensitivity as the Sagnac Interferometer ξ .
- It is therefore possible to assess the magnitude of the noise sources affecting LISA.
- This will allow us to measure a gravitational wave background in the frequency region of interest.